

Erratum

Drag Extrapolation to Higher Reynolds Number

Lance W. Traub

Embry-Riddle Aeronautical University, Prescott, Arizona 86301-3720

[*J. Aircraft*, 46(4), pp. 1458–1461 (2009)]

DOI: 10.2514/1.48011

Because of a resizing error in production, part of Fig. 4 was distorted. The correct version of this figure appears below.

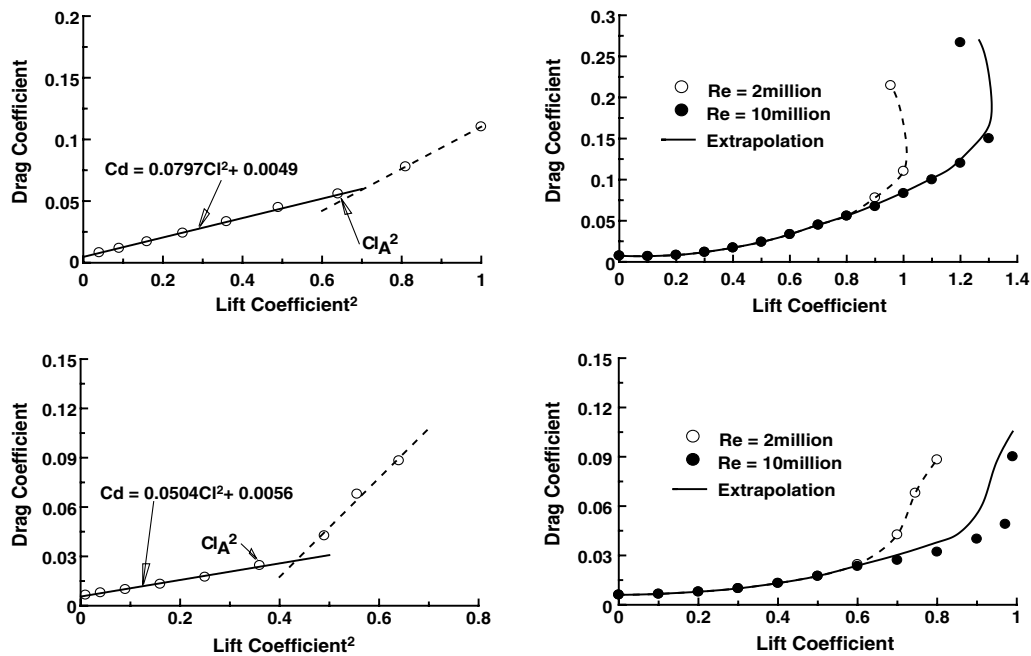


Fig. 4 Comparison of extrapolated polar with experiment: a) $AR = 5$, N64A312 section and b) $AR = 10$, N65A012 section. Data from [1].

Reference

- [1] Tinling, B., and Kolk, W. R., "The Effects of Mach Number and Reynolds Number on the Aerodynamic Characteristics of Several

12-Percent Thick Wings Having 35 Deg of Sweepback and Various Amounts of Camber," NACA RM A50K27, Feb. 1951.